

BGA SDMP 267 template (only for use in BGA airworthiness system)

Aircraft Maintenance Programme (AMP)								
	Aircraft identification							
1	Registration:		Type: ASK18		Serial no(s): 18002			
			Basis for the n	naintenance programn	ne			
2	Sailplanes, Self- Programme cor	Launching/Sustaining sai nplying with M.A.302(i)	lplanes and TMG ☑ Note the BGA	, not involved in commer SDMP 267 lists all the sch	ance program. This templat cial operations, declaring th neduled inspection requirer num Inspection Programme	ne "other" nents in ta		
		Des	ign approval ho	older (DAH) maintenanc	e data			
3	Equipment mar	nufacturer and type:		Applicable maintenanc	e data reference (at latest	revision)		
3a	Aircraft ALEXANDER SCHLEICHER OHG (other than balloons)			ASK18 Maintenance relate to this glider typ	Manual updated with all to be and serial number.	echnical r	notes that	
3b	Engine (if applicable)	N/A		N/A				
3c	Propeller (if applicable) N/A N/A							
Additional maintenance requirements not covered above (applicable to all AMPs, regardless of whethe design approval holder (DAH) data or minimum inspection programmes (MIPs))						er they a	re based on	
4	Indicate if any of the following additional maintenance requirements are applicable (when replying 'YES', list the specific requirements in Appendix B (add to the BGA SDMP 267 EASA mandatory and BGA CAMO requirements found after task 89) to this AMP							
	Maintenance du	ue to specific equipment	and modificatior	15			No	
	Maintenance due to life-limited components					Yes		
	Maintenance due to mandatory continuing-airworthiness information (airworthiness limitations (ALIs), certification maintenance requirements (CMRs), specific requirements in the TCDS, etc.)							
	Maintenance due to repetitive ADs					Yes		
	Maintenance due to specific operational/airspace directives/requirements (altimeter, compass, transponder, etc.)						No	
	Maintenance du	ue to type of operation o	r operational app	provals			No	
5			-		bulletins, service letters, Appendix B to this AMP)		No	

		Pilot-owner maintenance					
6	Does the Pilot-owner perform Pilot-owner m	aintenance (ref. Part-M)			No		
	If yes, enter the name of the pilot-owner(s):						
	Pilot-owner name	Licence Number:					
	Signature:	Date:					
	Approval/declara	tion of the maintenance programm	ne by owner		L		
7	Declaration by owner						
	'I hereby declare that this is the maintenance programme applicable to the aircraft referred to in Field 3, and I am fully responsible for its content and, in particular, for any deviations from the Design Approval Holder's recommendations.'						
	Signature/name/date:		2019				
		Certification statement					
8	'I will ensure that the aircraft is maintained in will be reviewed and updated as required.'	accordance with this maintenance pr	rogramme and that the i	maintenar	nce programme		
	Signed by the person/organisation responsible for the continuing airworthiness of the aircraft according to MLA.201:						
	Owner 🖾 — Lessee 🗌						
	Name of owner/lessee:						
	Address:						
	Telephone/fax:						
	Email:						
	Signature/date:						
9	Appendices attached to BGA SDMP 267						
	— Appendix A YES 🔀 NO 🗌 BGA :	SDMP 267 already complies with Apper	ndix A requirement				
	 Appendix B YES NO Add to the BGA SDMP 267 EASA mandatory and BGA CAMO requirements found after task 89 						
	Record of periodic reviews and revisions of the Aircraft Maintenance Programme (in accordance with M.A.302(g) or M.A.302(h)5, as applicable) (add more rows/lines if required)						
10	Entire below changes to this SDMP and revisi	on number	Date and sigr	nature of c	owner		
	Issue 1		/ /2019				



Work pack file ref:						
Page No:	Total pages in					
workpack						

BGA No:

Type. ASK18

Serial No: 18002

Task	ask Description Inspection detail		Operation Insp/check				
Item			itials				
asks 1	to 62 applicable to a	II aircraft (delete row/line as or write N/A as required)					
		ered sailplanes (delete row/line as or write N/A as required)					
	1						
	All Tasks General The aircraft must be clean. Inspect for security, damage, wear, integrity, drain/vent holes clear, signs of						
•		overheating, leaks, chafing, cleanliness and condition as appropriate to the particular task. Whilst					
0		checking GRP Composite structures check for signs of impact or pressure damage that may indicate					
		underlying damage.					
	Fucelage	The manufacturer's maintenance manual must be used for specific maintenance instructions.					
1	Fuselage Inspect external surface and fairings, gel coat, fabric, metal skins and paintwork. Check that						
	Paint/Gelcoat	registrations marks are correctly applied. All turbulator tapes are fitted correctly and secure.					
2	Fuselage	Check frames, formers, tubular structure, skin and attachments. Inspect for signs if corrosion on tubular framework.					
2	structure Nose Fairing						
3	Rudder	Inspect for evidence of impact with ground or objects. Inspect nose tow release unit and aperture. Check rudder assembly, hinges, attachments, balance weights.					
4	Pot	Check alignment of probe, check operation of ventilator and canopy demisting.	-				
5	Pitot/Ventilator	Check alignment of probe, check operation of ventilator and callopy demisting.					
	Centre section	Inspect for security, damage and condition.					
6	fairing						
	Wing	Inspect the wing structural attachments. Check for damage, wear and security. Check for rigging					
7	attachments	damage. Check condition and security of wing attachment pins and associated bearings.					
	Canopy, doors,	Inspect canopy/door and frame and transparencies for cracks, unacceptable distortion and					
	locks, jettison	discoloration. Check operation of all locks and catches.					
8	····,,,····	Carry out an operational test of the canopy jettison system from all positions.					
		Canopy jam during jettison inspection ref BGA Inspection 021/10/2001.					
		Check canopy/instrument panel gas strut inspection ref BGA Inspection 031/05/2002.					
	Seat / cockpit	Inspect seat (s). Check that all loose cushions are correctly installed and as appropriate, energy					
9	floor	absorbing foam cushions are fitted correctly and secured. Ensure that all seat adjusters fit and lock					
		correctly.					
	Cleanliness /	Check under cockpit floor/ seat pan and in rear fuselage for debris and foreign items.					
10	loose article						
	check						
11	Front skid/nose	Inspect for evidence of hard/heavy landings. Check skid wear. Inspect wheel, tyre and wheel box. Check					
	wheel & mounts	tyre pressure.					
	Mainwheel, tyre	There should be zero play (unless a tolerance is specified in the manual) in the brake torque link/stud.					
	& brake	Check for integrity of hydraulic seals and leaks in pipe work. Remove brake drums, check brake lining					
12	assembly	wear. Check disk/drum wear. Refit drum. Check brake adjustment.					
		CAUTION: BRAKE DUST MAY CONTAIN ASBESTOS.					
		Check operation of brake.					
		Tyres check for wear, sidewall damage, perishing, correct pressure and creep marks have not moved.					
42	Undercarriage	Check springs, bungees, shock absorbers, and attachments. Check for signs of damage.					
13	suspension	Service strut if applicable. If rubber parts fitted check for perished rubber and bulges.					
	l la de see suite see	Note: Carry out with weight off the landing gear.	N1/A				
14	Undercarriage	Check retraction mechanism and controls with aircraft on jacks/dolly, check warning system if fitted,	N/A				
14	retraction	gas struts, doors and linkages/springs, over centre/locking device. Perform retraction test.					
	system	Increase for avidence of bard /basis landings (back stid wars thereast wheel there and wheel have the					
15	Tail wheel	Inspect for evidence of hard/heavy landings. Check skid wear. Inspect wheel, tyre and wheel box. Check					
		bond of bonded skids. Check tyre pressure.					

	Release hooks	Inspect nose and C of G release hooks and controls as per manufacturer's instructions.			
16		Check operational life against manufacturer's instruction (both calendar life and actuations). Carry out operational test. If more than one release hook or control is fitted check operation of all release hooks from all positions.			
17	Harnesses	Inspect all harnesses for condition and wear of all fastenings, webbing and fittings. Check operation of release and adjustments. See BGA AMP manual Leaflet 4-8 for advice.			
18	Flight/rudder pedal assemblies	Inspect rudder pedal assemblies and adjusters.			
19	Rudder control circuit & stops	Inspect rudder control rods/cables. Check that control stops are contacting and secure. Pay attention to wear and security of liners and cables in "S" tubes. Check rudder assembly, hinges, attachments and balance weights are secure. Rudder cable wear (every 200FH or annual)			
20	Elevator control circuit & stops	Vith the tailplane derigged, check tailplane attachments, inspect elevator control rods/cables. Check hat control stops are contacting and secure. Inspect self-connecting control devices, check gel coat, abric covering or metal skin.			
21	Aileron and flap control circuit & stops	Inspect aileron control rods/cables. Check that control stops are contacting and secure. Inspect self-connecting control devices.			
22	Flap control circuit and detents	Inspect flap control circuit, check any gas struts fitted work as specified. Check that all detents and springs in the flap circuit and handle operate correctly as specified by Manufacture and detents are not too excessively worn.	N/A		
23	Trimmer control circuit	Inspect trimmer control rods/cables. Check friction/locking/connecting devices.			
24	Air brake control circuit	Inspect air brake control rods/cables/bellcranks/bracket. Check friction/locking device (if fitted). Inspect self-connecting control devices.			
25	Wheel brake control circuit	Inspect wheel brake control rods/cables. If combined with air brake, ensure correct rigging relationship and you can still achieve full airbrake. Check parking brake operation (if fitted).			
26	Instrument panel assemblies	Inspect instrument panel and all instruments/equipment. Check that instrument readings are consistent with ambient conditions. Check marking of all switches, circuit breakers and fuses are correctly labelled. Registration is displayed on instrument panel. Check operation of all installed equipment as possible i.a.w. manufacturer's instructions. Check all instruments are marked as required by Flight Manual.			
27	Pitot/static system	Inspect pitot probes, static ports all tubing (as accessible) for security, damage, cleanliness, kinking and condition. Drain any water from condensate drains. Perform system leak check.			
28	ASI operational check	Carry out accuracy of the airspeed indicator (in situ permissible) i.a.w. manufacturer's instructions (use manufacturers limits. If Not avail. Max error 2 knots). Ensure colour coding has been applied if required in flight manual.			
29	Altimeter datum	Check barometric sub scale. (max. error 2 Mb).			
30	Electrical installation/ fuses/trips	Check all electrical wiring for condition. Check for signs of overheating and poor connections. Check fuses/trips for condition and correct rating.			
31	Battery	Check battery mounting for security and operation of clamp. Check for evidence of electrolyte spillage and corrosion. Check that battery has the correct main fuse fitted. It is recommended to carry out battery capacity test on gliders equipped with radio, used for cross- country, airways or competition flying. Note: In accordance with equipment manufacturer's recommendations where capacity checks are recommended by the equipment manufacturer. See BGA AMP manual leaflet 4-9.			
32	Oxygen systems	Inspect oxygen system. Check bottle hydrostatic test date expiry i.a.w. manufacturer's recommendations. Ensure that bottle is not completely empty (200psi min) refill with aviator's oxygen only. Clean masks and regulators with approved cleaning wipes. Ensure that oxygen installation is recorded on weight and C of G schedule. Check all instruments are marked as required by Flight Manual. CAUTION: OBSERVE ALL SAFETY PRECAUTIONS	N/A		
33	Radio installations and placards, transponders	Check radio installation, microphones, speakers and intercom if fitted. Check that call sign placard is installed. Check aircraft registration placard is visible near radio. Carry out radio ground function test. Record type fitted. All avionics (including transponders) to be maintained as per the manufacturer's instructions and applicable ADs.			
34	Radio frequency check	48-month frequency tolerance check. (Not required for modern 720/760 or later channel transceivers)	N/A		
35	Removable ballast	Check removable ballast mountings and securing devices for condition. Check that ballast weights are painted a conspicuous colour. Check that provision is made for the ballast on the loading placard. Check that the ballast arrangements as configured are supported by the Flight Manual (technical notes often require flight manual amendments).			

	Colour coding of controls	Ensure that controls are colour coded and in good condition, as follows; Tow release: Yellow				
36		Air Brakes: Blue Trimmer: Green Canopy normal operation: White				
		Canopy jettison: Red Combined Canopy jettison and normal operation: White and Red Other controls: clearly marked but not using any of the above colours.				
37	Equipment stowed in centre section	Check for security and condition. Check validity of any safety equipment. Check manufacturer's and NAA (if required) data plates.				
38	Wing struts/wires	spect struts for damage and internal corrosion. Re-inhibit struts internally every 3 years or in cordance with manufacturer's instructions.				
39	Drag chutes & controls	Check for correct operation. Inspect chute, rigging lines, packing and release mechanism. Check				
40	Water ballast system	 repackaging date. Check water ballast system, wing and tail tanks as fitted. Check filling points, level indicators, vents, dump and frost drains for operation and leakage. If loose bladders are used check for leakage and expiry date as applicable. Ensure outside temp gauge is fitted and reads ambient temperature. 				
41	Tailplane and elevator	With tailplane de-rigged check tailplane and attachments, self-connecting and manual control connections, check condition of gel coat, fabric or metal skin. All turbulator tapes are fitted correctly and in secure. Check condition and fitment of sealing tape ref BGA Inspection 009/10/2000. Control tape and Mylar seal inspection ref BGA Inspection 011/12/2000.				
42	Left wing	Wooden structure inspection ref BGA Inspection 047/02/2006.Check mainplane structure externally and internally as far as possible. All vents and drain holes are clear. Check gel coat or fabric covering. Check registration marks are correctly applied. Ensure all boundary layer blow holes are not blocked and pressure feed system for them is serviceable. All turbulator tapes are fitted correctly and secure. Ensure compliance with Generic Requirement 8, Fabric Inspection.				
43	Left wing controls	Inspect aileron and flaperon assemblies, hinges, control connections, springs/bungees, tapes and seals. Ensure that seals do not impair full range of movement. Control tape and Mylar seal inspection ref BGA Inspection 011/12/2000.				
44	Left air brake/spoiler	Inspect air brake/spoiler panel(s) operating rods, closure springs, and friction devices as fitted. Check locking forces if specified by manufacturer or AD.				
45	Left Flap	Check flap system and control. Inspect self-connecting control devices.	N/A			
46	Right wing	Check main plane structure externally and internally as far as possible. All vents and drain holes are clear. Check gel coat or fabric covering. Check registration marks are correctly applied. Ensure all boundary layer blow holes are not blocked and pressure feed system for them is serviceable. All turbulator tapes are fitted correctly and secure. Ensure compliance with Generic Requirement 8, Fabric Inspection. Wooden structure inspection ref BGA Inspection 047/02/2006 (5 year repeat).				
47	Right wing controls	Inspect aileron and flaperon assemblies, hinges, control connections, springs/bungees, tapes and seals. Ensure that seals do not impair full range of movement. Control tape and Mylar seal inspection ref BGA Inspection 011/12/2000.				
48	Right air brake/spoiler	Inspect air brake/spoiler panel(s) operating rods, closure springs, and friction devices as fitted. Check locking forces as specified by AMM or AD				
49	Right Flap	Check flap system and control. Inspect self-connecting control devices.	N/A			
50	Bonding/vents/ drain	Check all bonding leads & straps. Check all vents and drains are clear from debris.				
51	Lubrication	Lubricate and replenish fluids in accordance with manufacturer's requirements				
52	Markings	Check side and under-wing markings are correct. If applicable, an exemption for alternate display is approved. Ident plate for CAA registered aircraft present. Identification plate for National Aviation Authority registered aircraft is present. Other identification markings in accordance with local (national) rules. BGA Number on fuselage for BGA registered aircraft.				
53	Mandatory checks	Check for compliance of all mandatory modifications, airworthiness directives and inspections applicable to the airframe, accessories & equipment. Record compliance in the logbook. State of design Type Certificate and STC holder AD list, BGA Compendium, BGA Technical News Sheet, BGA Mandatory inspections, manufacturer's mandatory check list (if available).				
54	Manufacturer's recommendation and life inspections	Review manufacturer's maintenance schedules and instructions for continued airworthiness for the airframe to establish if any additional work, servicing or preservation action is required. Any Deviations from TCDS holder's recommendations must be recorded and signed for by the owner.				
55	Control deflections & free play	Check and record range of movements and cable tensions (if specified), check free play. (Record in Workpack)				
56	Duplicate inspections	Record each item requiring a duplicate inspection on an additional worksheet and complete prior to releasing aircraft back to service.				

57	maximum landing mass or, for aeroplanes, the cumulative change in CG position exceeds 0.5 % of the mean aerodynamic chord. This may be done by weighing the aircraft or by calculation. If the AFM requires to record changes to mass and CG position below these thresholds, or to record changes in any case, and make them known to the pilot-in-command, mass and CG position should be revised accordingly and made known to the pilot-in-command.					
58	Speed/weight/ manoeuvre placard	Check placard is correct and legible and accurately reflects the status of the aircraft.				
59	Hours	Hours at this inspection:				
60	Launches	Launches at this inspection:				
61	Modifications	Review Log Book and verify that any modifications incorporated since last Airworthiness Certificate or ARC renewal have been approved and correctly embodied and recorded.				
62	Log book	Complete log book entry. Ensure that all flying records are entered and up to date.				
63	Flight Manual	Verify that the Aircraft Flight Manual or Operating Handbook is at the latest revision.				
Tasks 63	3 to 89 are only appli	cable to Powered Sailplanes				
6 4	64 Engine pylons & mountings & inspect engine and pylon installation. Check engine compartment and fire sealing. Check pylon cracks and delamination if made from composites. Ensure all rubber parts (especially engine mounting are not perished, cracked or deteriorated. Check starter motor security, casing, wiring, condition drive gear and flywheel if fitted. 64 Engine pylons & mountings & inspect engine and pylon installation. Check engine compartment and fire sealing. Check pylon cracks and delamination if made from composites. Ensure all rubber parts (especially engine mount in the perished, cracked or deteriorated. Check starter motor security, casing, wiring, condition drive gear and flywheel if fitted.					
65	Gas strut	Check gas strut with AMM.	N/A			
66	Pylon/engine stops	Check limit stops on retractable pylons. Check restraint cables.	N/A			
67	Electric actuator	Inspect electric actuator, motor, spindle drive and mountings.	N/A			
68	Electrical wiring, Inspect all electrical wiring. Pay special attention to wiring that is subject to bending during extension external and and retraction of engine/pylon. internal Check function of all lights. lights/strobes/ beacons					
69	Limit switches	Check operation of all limit switches & strike plates. Ensure not damaged by impact.	N/A			
70	Fuel tank	Check fuel tank mountings, electrical bonding and tank integrity. Check fuel quantity indication system if fitted. If a GRP tank is fitted ensure the integrity of the internal resin in case it has been affected by etheral and other parts is a set of in case it has been affected by	N/A			
71	Fuel pipes & vents	engine/pylon. Check vents clear. Ensure overboard drains do not drain into engine compartment. Check self-sealing couplings. Ensure all swaged fittings, jubilee clips are secure and there is no				
72	Fuel cock or SOV	perishing. Check operation of fuel cock or shut off valve & indications.	N/A			
	Fuel pumps and	Clean or replace filters as recommended by manufacturer	N/A			
73	filters	Check operation of fuel pumps for engine supply or tank replenishment. Check fuel pump controls and indications.				
74	Decompression valve	Inspect decompression valve and operating control.	N/A			
75	Spark plugs	Carry out spark plug service. It is recommended to replace spark plugs at annual intervals.	N/A			
76	Harnesses & magneto	Inspect low tension and high-tension wiring, connectors, spark plug caps. Check magneto to engine timing. Check impulse coupling operation.	N/A			
77	Propeller	Inspect propeller, hub, prop bolts torque (if require) folding mechanism, brake, pitch change mechanism, stow sensors, belts and pulleys condition and tension. Lubricate all as required by TCDS holder. Check overhaul period and TBO of propeller.	N/A			
78	Doors	Check engine compartment doors, operating cables, rods and cams.	N/A			
79	Safety springs	Check all safety and counterbalance springs.	N/A			
80	Extension and retraction	Check extension and retraction operation times are within limits specified by manufacturer. Check light indications and interlocks for correct operation. Check for factory software updates every year.	N/A			
	Exhaust.	Inspect exhaust system, silencer, shock mounts and links. Pressure test cabin and carb heater exhaust heat exchanger (if applicable). Check turbocharger as required by TCDS holder. See CAA CAP 562 CAAIP Leaflet B-190 for further guidance				
81	turbocharger, cabin and carburettor heat.					
81 82	turbocharger, cabin and		N/A			

	Engine	Inspect all engine instruments and controls. Check control unit, mounts, bonding and connections.	N/A				
84	instruments and	Carry out internal self-test if fitted. Check engine and propeller controls for full and free movement –					
	controls	throttle, mixture, carburettor heat, cowl flaps and propeller pitch.					
05	Engine battery	(if separate to airframe battery) Inspect battery and mountings. If main fuse is fitted check rating and	N/A				
85		ondition. Carry out capacity test, refer to AMM I for guidance.					
86	Placards	Check all placards in accordance with Flight/Maintenance Manual and are legible.	N/A				
	Oil and fuel leaks	Perform ground run (except with dive start engines). Check temperatures and pressures and indication	N/A				
87		within permitted range. With the engine fully serviced (and ideally still warm from a check run) check					
		the fuel and oil system for leaks.					
	Mandatory	Check for compliance of all mandatory modifications, Airworthiness Directives and inspections	N/A				
	checks	applicable to the engine, propeller, accessories & equipment. Record compliance in the logbook.					
00		TCDS holder AD list, EASA AD list, Equipment ADs (including Technical notes and Service Bulletins) BGA					
88		Compendium, BGA Technical News Sheet, BGA Mandatory Inspections, BGA Compendium, in service					
		issues, manufacturer's mandatory check list (if available) and factory service bulletins and technical					
		notes.					
	Manufacturer's	Review manufacturer's maintenance schedules and instructions for continued airworthiness for the	N/A				
89	recommendation	engine/propeller to establish if any additional work is required. All recommendations not carried out					
		require an owner declared deviation.					

EASA Mandatory item are recurring (add more	s. Add ALIs (found in section 4 of modern AMM and TCDS), only add EASA and State of Desi rows/lines if required)	gn ADs that
LBA AD1989-018/3	TOST Hook condition and life (mandatory 10000 actuations and recommended four-year life)	10000 ACTUATIO NS/4- YEARS
AD 72-7/3 (BGA 043/07/2004)	ELEVATOR ROOT RIB INSPECTION	ANNUAL
TN8	ELEVATOR CENTRAL SUPPORT INSPECTION (CLUB DECISION TO REPORT ANNUALLY	ANNUAL
recommendations (i	irements (found in Compendium and BGA inspections) and if desired add advisory Maintenau f embodied and not already included in the SDMP 267) add more rows/lines below if required. Intenance you want to include on this form. For instance, Flarm software updates or reminders Maintenance Manual.	You can also
TNS 011/12/200 TNS 12/00 TNS 5-2010	INSPECTION OF CONTROL TAPES AND SEALS	ANNUAL
TNS 05/2013	ON CONDITION OPERATION OF HARNESS/STRAPS IF INEXCESS OF OEM LIFE	ANNUAL
TNS 04/2014 & BGA Mandatory Inspection 056-08	CONTROL GRIPS – ENSURE SECURE AND NOT ABLE TO TWIST	ANNUAL
BGA COMPENDIUM	RE-WEIGH AT LEAST EVERY 8 YEARS	8 YEARS
042/07/2004 STRUCTURAL INSPECTIONS	WAS PREVIOUSLY EVERY FIVE YEARS, NOW EVERY THREE YEARS). WING AND FIN INSPECTIONS	EVERY 3 YEARS
		NEXT DUE NLT 27 MAY 21
BGA 043/07/2004 ISSUE 2	ELEVATOR ROOT RIB 1 INSPECTION IAW BGA INSTRUCTION	ANNUAL
2013-0091 (BGA 010/12/2000)	ELEVATOR DRIVE (PREVIOUSLY TN9) – CHECK OF ALIGNEMENT	ANNUAL
FLARM and Airspace Update	UPDATE FLARM AND AIRSPACE SOFTWARE	ANNUAL

Add any Deviations from TCDS holder and equipment manufacturer recommendations from mandatory service bulletins, AMM, AFM and TCDS. The BGA requires justification and Acceptable Means of Compliance for Deviations. No deviations are permitted from Airworthiness Directives or mandatory maintenance (ALIs) or BGA CAMO requirements as specified in the Maintenance/Flight Manuals, TDCS, ADs and BGA Compendium (add more rows/lines if required)				
Service/life/TBO Inte	erval	Task Description	Engineering justification and alternative means of compliance (AMC).	
TC holder recommendations (hrs/cyc/cal)	Changed to		Add extra documents to this MIP section as required to support AMC and engineering justification of a deviation.	
12 YEARS	ANNUAL REVIEW	GADRINGER HARNESS LIFE	HARNESS LIFE EXTENDED FROM 12 YEARS TO ANNUAL REVIEW SUBJECT TO ANNUAL INSPECTION USING BGA AMP 4-8 GUIDELINES. WITHIN THE BGA CAMO STRICT ADHERENCE TO BGA AMP 4-8 GUIDELINES HAS SHOWN SEAT HARNESSES HAVE BEEN SAFELY EXTENDED WITH NO LOSS/PERCEIVABLE LOSS OF SERVICEABILITY. REFER TO TNS 02/05 & 03/2009	
4 YEARS	3000 LAUNCHES	(TOST SAFETY RELEASE MECHANISM EUROPA G72 & G73 JAN 1989 OR G88 OF FEB 1989, TOST SAFETY RELEASE MECHANISM EUROPA E72 & E75 MARCH 1989 OR E85 MARCH 1989.) ALL TOST RELEASE UNITS ARE SUBJECT TO AD 1989-018/3, TN 1-2001	BGA GLIDERS MAY USE 4-5 ACTUATIONS PER FLIGHT FOR CLUB GLIDERS = 2000 TO 2500 LAUNCHES. PRIVATE GLIDERS MAY USE 3-4 ACTUATIONS PER FLIGHT = 2500 TO 3000 LAUNCHES. REFER TO TNS 02/05	

General Remarks						
Date of ARC or BGA C of A expiry:						
Other remarks:						
Record identifying marks.	Fin:	Fuselage:	Under wing:			
Certificate of Release to Service						
All work has been recorded in the ap	propriate logbook and all additiona	I worksheets have accounted for a	nd certified and for BGA registered			
gliders.						
EASA Aircraft - Certifies that the	work specified, except as otherwis	e specified, was carried out in acco	ordance with Part-M and in that			
respect is considered ready for release to service. BGA Approval No. UK.MF.0007.						
(* Written signature required)						
Inspector Name:	Signed	Date:	BGA Inspector No:			